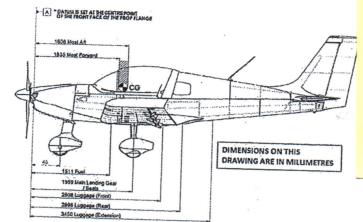
## SLING AIRCRAFT (PTY) LTD MASS AND BALANCE REPORT

AIRCRAFT TYPE : SERIAL NO. :

Sling LSA 340

**REGISTRATION:** N56RRC Empty weight = 906.1 lbsEmpty 
Moment = 60541.11 lbs. in



The method of calculation of the aircraft empty mass, total aircraft moment, centre of gravity and percentage mean aerodynamic chord appear from formulae set out in the table and the formulae below.

(Comply with the provisions of the aircraft maintenance manual when performing aircraft empty mass and balance)

|   |                     | ltem              | Weight [kg]      |          | Arm formal |                        |         |                        |     |              |
|---|---------------------|-------------------|------------------|----------|------------|------------------------|---------|------------------------|-----|--------------|
|   | Nircarft<br>npty CG | Nose Wheel        | 111              |          | Arm [mm]   |                        |         | Moment [kg.mm]         |     |              |
|   |                     | Left Main Wheel   |                  | 72.0 kg  |            | =                      | 464 mm  | $M_N = W_N \times L_N$ | =   | 33408 kg.mm  |
|   |                     |                   | W <sub>L</sub> = | 171.0 kg | Lį         | =                      | 1959 mm | $M_L = W_L \times L_L$ |     |              |
|   |                     | Right Main Wheel  | W <sub>R</sub> = | 168.0 kg | Lo         | =                      | 1959 mm |                        |     | 334989 kg,mm |
|   |                     | Computed CG Empty | Empty weight     | _        |            |                        |         | $M_R = W_R \times L_R$ |     | 329112 kg.mm |
|   |                     |                   | W <sub>E</sub> ≈ | 1        |            | Total aircraft moment: |         |                        | it: |              |
| , |                     |                   | AAE =            | 411.0 kg |            |                        |         | M <sub>T</sub>         | =   | 697509 kg.mm |

Aircraft CG = 
$$\frac{\text{Total aircraft moment}}{\text{Aircraft empty weight}} = \frac{M_T}{W_E} = \frac{697509}{411.0} = 1697 \, \text{mm}$$

CG as percentage of MAC =  $\frac{(\text{CG} - 1366)}{1339} \times 100 = \frac{1697 - 1366}{1339} \times 100 = 24.7 \%$ 
I hereby certify that the information as recorded above is correct in every respect to the best of my knowledge

Name: SEAN RUSSELL

KING AIRCRAL Certifying Inspector 08

Date: 20/01/2022

Signature:

